

Composite Panel Checklist

TEST SPECIMEN #

Please Indicate Type of Testing Preferred:

Streamlined testing (FAA 8130-9 required)

Test plan requested

Company Name:

Contact Name:

Phone:

E-Mail:

Date Sent:

P.O. #:

Aircraft Make:

Model:

S/N #:

Tail #:

*What area of the aircraft does this panel represent?
(i.e. sidewall, cabin floor, crew seat, etc.)*

TEST DATA IS IN SUPPORT OF (As Required):

Field Approval

Major Repair or Alteration

Minor Repair or Alteration

Organization Designation Authorization (ODA)

Type Certificate (TC)

Technical Standard Order (TSO)

Supplemental Type Certificate (STC)

FAA form 8110-3 may only be issued aircraft specific for U.S. registered or U.S. State of Design aircraft.

Please complete checklist for each panel submitted. A completed FAA form 8130-9, test specimen drawings, and copies of packing slips or invoices providing traceability for each panel component (Nomex, aluminum, foam, veneer, leather, fabric, adhesive, etc.) must be included for each panel set submitted.

TYPES OF COMPOSITE TESTS AVAILABLE:

1. Horizontal 23.853 2. 12-Second Vertical 25.853(a) 3. 60-Second Vertical 25.853(a)
4. 45-Degree Angle 25.855(d) App F Part 1 5. 60-Degree Wire 25.869(a) App F Part 1
6. Applicable Tests for the "as installed state" for this aircraft

If Skandia is fabricating the test specimens, does Skandia have your permission to issue FAA 8130-9 forms on your behalf? Yes No

REQUIREMENTS FOR SPECIMENS TO BE TESTED:

Horizontals, 12- and 60-Second Verticals less than .25": Woven specimens (fabric, carpet, etc.) three strips, 3" x 12" cut from across the roll **AND** three strips, 3" x 12" cut from up the roll. Non-woven specimens (leather, vinyl, plastic, foam, etc.) three strips, 3" x 12". **12- and 60-Second Verticals more than .25":** Woven specimens (fabric, carpet, etc.) six strips, 3" x 12" cut from across the roll **AND** six strips, 3" x 12" cut from up the roll. Non-woven specimens (leather, vinyl, plastic, foam, etc.) six strips, 3" x 12". **45-Degree Angle:** Three specimens, 8½" x 8½". **60-Degree Wire:** Three specimens, 34" long.

Per FAA Handbook, DOT/FAA/AR-00/12 Chapter 1.0, when testing two samples per 1.6.2.4 for test specimens greater than 0.25" (6.35 mm), the recorded burn length will be from the face with the greatest burn length in accordance with section 1.6.2.9

If edge filler is used, it must be called out on the 8130-9 but should be omitted from both the material list and the composite sketch as it will be eliminated from the test specimens per 14 CFR 25.853 App F Part 1(b)(2).

As required by App F, foam shall be tested in 1/2" thicknesses. If the panel consists of two or more foams glued together, the foam specimens should be two 1/4" (three 1/16", etc.) pieces glued together per AC 25-17 Transport Airplane Cabin Interiors Crashworthiness Handbook or other accepted FAA test method.

Per FAA Fire Handbook, DOT/FAA/AR-00/12, Chapter 1, Paragraph 1.4.4.1 "For testing purpose where a part is used in several thicknesses, the minimum thickness will be tested."

Additional reference material: Advisory Circular AC 25—853-1, AC 21-25A, AC23-2 and Aircraft Materials Fire Test Handbook DOT/FAA/AR-00/12, FAA Policy Statement PS-ANM-25.853-01.

IF USING A PROCESS SPECIFICATION CALLOUT, A COPY OF THE PROCESS SPECIFICATION MUST BE INCLUDED.

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